Development of a two-phase compressible Euler numerical solver with applications to shock wave interactions with gas bubbles

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Two Phase Compressible Euler Solver

In each phase solve the compressible Euler equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot \left(\rho \underline{u} \right) = 0$$

$$\frac{\partial}{\partial t}(\rho \underline{u}) + \nabla \cdot \left(\rho \underline{u} \underline{u} + p \underline{I}\right) = 0$$

$$\frac{\partial \mathbf{E}}{\partial t} + \nabla \cdot \left(\underline{\mathbf{u}} (\mathbf{E} + \mathbf{p}) \right) = \mathbf{0}$$

$$\mathbf{E} = \rho \mathbf{e} + \frac{1}{2} \rho \left| \underline{\mathbf{u}} \right|^2$$

$$p = B \left(\frac{\rho}{\rho_0}\right)^{\gamma} - B + A$$

 $\mathbf{p} = (\gamma - 1) \rho \mathbf{e}$



Conservative Form

$$\frac{\partial \underline{U}}{\partial t} + \frac{\partial \underline{F}(\underline{U})}{\partial x} + \frac{\partial \underline{G}(\underline{U})}{\partial y} + \frac{\partial \underline{H}(\underline{U})}{\partial z} = 0; \qquad \underline{U} = (\rho, \rho u, \rho v, \rho w, E)^{T}$$
$$\underline{F}(\underline{U}) = \begin{pmatrix} \rho u \\ \rho u^{2} + p \\ \rho u v \\ \rho u w \\ (E+p)u \end{pmatrix}, \qquad \underline{G}(\underline{U}) = \begin{pmatrix} \rho v \\ \rho u v \\ \rho v^{2} + p \\ \rho v w \\ (E+p)v \end{pmatrix}, \qquad \underline{H}(\underline{U}) = \begin{pmatrix} \rho w \\ \rho u w \\ \rho v w \\ \rho v w \\ \rho w^{2} + p \\ (E+p)w \end{pmatrix}$$



Compressible Euler Solver in each Phase

- Method of lines applied to separate spatial and temporal integration
- Spatial fluxes 3rd order ENO-ROE method employed with similarities to the Marquina flux splitting technique
- In time a 3rd order TVD Runge-Kutta method employed
- Timesteps calculated adaptively to ensure CFL condition meet
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Interface Evolution

Interface captured using a level set function $\phi(\underline{r}, t)$

$$\frac{\partial \phi}{\partial t} + \mathbf{v} \cdot \nabla \phi = 0$$

Reinitialise to a sign distance function

$$\frac{\partial \varphi}{\partial \tau} + \hat{S}(\varphi_0) [\nabla \varphi | -1] = 0$$



Original Ghost Fluid Method

- Gas/Gas Systems
- Pressure and velocity from real fluid of phase 1 copied into the ghost fluid of phase 2 and vice versa
- Density extrapolated from real fluid into ghost fluid for a given phase

$$\frac{\partial I}{\partial \tau} + \underline{n} \cdot \nabla I = 0$$



Modified Ghost Fluid Method

- Air pressures and water normal velocities dictate interface values
- Gives greater dissipation at the interface, thus offering greater stability in order to counter numerical issues associated with the stiff nature of the Tait equation



Ghost Air Cells

- Copy real values of the normal velocity components of water into each node in the ghost region;
- Extrapolate pressure & tangential components of velocity from the first set of cells adjacent to the interface in the real air region across the interface;
- Set density values using the equation $\rho = \hat{\rho} \left(\frac{p_0}{\hat{p}} \right)^{1/\gamma}$ where $\hat{\rho}$, \hat{p} obtained by extrapolating the density and pressure values from the second set of cells relative to the interface in the real air region ∇

Ghost Water Cells

- Copy real air pressure and tangential velocity values into each node in the ghost water region;
- Extrapolate normal components of velocity from the first set of cells adjacent to the interface in the real water region;
- Set density values using the equation $\rho = \hat{\rho} \left(\frac{p_0 + B_2}{\hat{p} + B_2} \right)^{1/\gamma}$

where $\hat{\rho}$, \hat{p} obtained by extrapolating the density and pressure values from the second set of cells relative to the interface in the real water region





Impact of a krypton bubble in air by a shock wave



Pressure contours

Axisymmetry geometry

Initial radius 20mm

Shock strength 0.15698MPa

401x1601 grid

1st order ENO-ROE









6 air bubbles in water





3D Cartesian geometry





Issues

- accuracy of interface conditions
- conservation of mass
- numerical instability for strong shocks
- effective parallelisation
- extension of applied interface conditions to include phase change or chemical reactions



Modified Ghost Fluid Method

- A Riemann problem is defined and solved approximately to predict the interface state
- The predicted interface state is used to define ghost fluid states by interpolation.

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(a) 1201x1201 grid, *t*=6.0μs;
(b) 1201x1201 grid, *t*=6.6μs;
(c) 401x401 grid, *t*=6.0μs;
(d) 801x801 grid, *t*=6.0μs.



Conclusions

Much still to be done!!!



References

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